

Homework (Week - 15) Solutions !

Ch - 13

Problem 15:

$$(a) \quad F_g = \frac{G m_b m_f}{r^2} = \frac{6.7 \times 10^{-11} \times 4.20 \times 10^0}{(0.2)^2}$$

$$= 7.0349 \times 10^{-7} N$$

$$(b) \quad F_g = \frac{G m_b m_J}{r^2} = \frac{(6.7 \times 10^{-11}) (4.20) (1.898 \times 10^{27})}{(6.29 \times 10^{11})^2}$$

$$= 1.349 \times 10^{-6} N$$

Gravitational forces from baby's father and other objects added up would be roughly equal to force exerted on the baby by jupiter.

Problem 20:

$$(a) g = \frac{GM_e}{r^2}$$

$$M_e = \frac{gr^2}{G} = \frac{9.832 \times (6.356 \times 10^6)^2}{6.7 \times 10^{-11}} = 5.928 \times 10^{24}$$

M_e from NASA fact sheet is within 0.74% error of the mass calculated above.

Problem 32:

$$U_i = -\frac{G m_1 m_2}{r_i} = \frac{(6.7 \times 10^{-11})(25)}{0.15} = -1.116 \times 10^{-8} \text{ J}$$

Upon impact their centre-to-centre distance will be 10.2 cm.

$$K_i = 0 \quad K_f = \left(\frac{1}{2}mv^2\right)(2)$$

Upon impact \rightarrow They both will have the same velocity magnitude

$$K_f + U_f = K_i + U_i \quad r_i = 0.15 \text{ m} \\ r_f = 0.102 \text{ m}$$

$$mv^2 = U_i - U_f \quad m_1 = m_2 = m$$

$$v^2 = -\frac{Gm_1m_2}{r_i} + \frac{Gm_1m_2}{r_f}$$

Problem 40:

Using Kepler's third law.

$$T^2 = \frac{4\pi^2}{GM_s} r^3$$

$$M_s = \frac{4\pi^2}{GT^2} r^3 \quad r = 1 \text{ AU}$$

Problem 46:

$$T = 1 \text{ day} = (24 \times 3600) \text{ sec}$$

$$T^2 = \frac{4\pi^2}{GM_E} r_{\text{orbit}}^3$$

$$r_{\text{orbit}} = \sqrt[3]{\frac{GM_E T^2}{4\pi^2}}$$

Problem 64:

(a) Earth's rotation should be taken into account for determining the real weight.

At the pole, Earth's rotation doesn't affect your weight. In other words it is your

true weight due to Earth's gravitational pull. At the equator using the net centripetal force experienced by a mass, your weight is

$$mg - \frac{mv^2}{r} = W \Rightarrow W = m\left(g - \frac{v^2}{r}\right) = m\left(g - \omega^2 r\right)$$

$$W_{\text{equator}} = m\left(g - \omega^2 R_E\right) \quad \text{where}$$

$r = R_E$ at the equator.

$r = 0$ at the pole.

$$W_{\text{pole}} = mg$$

$$W_{\text{equator}} = \frac{1}{2} W_{\text{pole}}$$

$$\gamma h(g - \omega^2 R_{\text{planet}}) = \frac{1}{2} \gamma h g.$$

$$g - \frac{1}{2} g = \omega^2 R_E.$$

$$\omega^2 = \frac{\frac{1}{2} g}{R_E} = \frac{\frac{1}{2} (9.81)}{6.4 \times 10^6}$$

$$\left(\frac{T}{T}\right)^{-1} = \frac{(4\pi^2)(\frac{1}{2}g)}{R_E}$$

$$\left(\frac{T}{T}\right)^{-1} = \frac{(2\pi^2)g \cdot \frac{1}{2}}{R_E}$$

$$\left(\frac{2\pi}{T}\right)^2 = \frac{\frac{1}{2} g}{R_E}$$

$$\frac{2(4\pi^2)R_E}{g} = T^2.$$

You don't need to consider the shape.

Ch- 15:

Problem 21:

$$x(t) = A \sin(\omega t + \phi)$$

$$\frac{d^2x}{dt^2} = -A\omega^2 \sin(\omega t + \phi)$$

$$\ddot{x} = -\omega^2 x$$

$$a = -\omega^2 x \quad (\text{SHM})$$

Cosine function is chosen since we have the mass and spring having an initial displacement which maximum at $t = 0$.

$$x(t) = A \cos(\omega t + \phi)$$

$$\Rightarrow \frac{d^2 x}{dt^2} = -\omega^2 x \quad \text{same period of oscillations}$$

Problem 27:

Each piston has a frequency of $\frac{750}{8}$ sound so, after two revolutions piston makes a sound of frequency $\frac{750}{8} = 93.75 \text{ Hz}$.

$$\omega = (93.75)(2) \text{ rev/s}$$

$$\begin{aligned} \text{Speed of the car is } V &= (93.75)(2)\left(\frac{1000}{2000}\right) \\ &= \omega r \\ &= 93.75 \text{ m/s.} \end{aligned}$$

Problem 30:

$$\text{Use } T = 2\pi \sqrt{\frac{m}{k}} \quad k = \frac{4\pi^2 m}{(1.50)^2}$$

$$m = 0.5, T = 1.50 \Rightarrow k = \frac{4\pi^2 m}{(1.50)^2}$$

Now if $T = 2 \text{ sec}$

$$2 = \frac{2\pi}{\sqrt{\frac{m'}{4\pi^2 m / (1.50)^2}}}$$
$$= \frac{2\pi}{\sqrt{\frac{m' (1.50)^2}{4\pi^2 m}}}.$$

$$2 = \frac{2\pi}{2\pi} \cdot (1.50) \sqrt{\frac{m'}{m}}.$$

$$\left(\frac{4}{3}\right)^2 = \frac{m'}{m}.$$

$$m' = \frac{16}{9} m$$

$$m' = \frac{16}{9} (0.5) = 0.88 \text{ kg.}$$

We need to add a mass of 0.38 kg to make the period equal to 2 secs.

$$(d) F_c = \frac{GMm}{r^2}$$

$$\frac{mv^2}{r} = \frac{GMm}{r^2}$$

$$v^2 = \frac{GM}{r}$$

$$v = \sqrt{\frac{GM_{\text{asteroid}}}{r}} = \sqrt{\frac{(6.67 \times 10^{-11})(1.1 \times 10^{16})}{(8.75 + 5.40) \times 10^3}}$$

$$(e) V_{\text{esc}} = \sqrt{\frac{2GM_{\text{asteroid}}}{r_{\text{asteroid}}}}$$

$$= \sqrt{\frac{2(6.67 \times 10^{-11})(1.1 \times 10^{16})}{(8.75 \times 10^3)}}$$

Additional Problem:

(a) $W_{\text{mars}} = m g_{\text{mars}}$

$$g_{\text{mars}} = 3.72076 \text{ m/s}^2$$

(b) $E = \frac{1}{2}mv^2 + U_g$. where $U_g = 0$ for $r \rightarrow \infty$.

$$U_g = -\frac{GMm}{R_{\text{mars}}}$$

$$E_i = \frac{1}{2}mv_i^2 + \frac{GMm}{R_{\text{mars}}}$$

(c) For escaping. we assume $E_f = \frac{1}{2}mv_{\text{esc}}^2 + U_g^f$.

$$U_g = 0$$

$$E_i = E_f$$

$$\frac{1}{2}mv_{\text{esc}}^2 = \frac{GMm}{R_{\text{mars}}}$$

$$v_{\text{esc}} = \sqrt{\frac{2GM}{R_{\text{mars}}}}$$